



INSPECTION CHECKS

AUTO-PILOT (YAW) INSTALLATIONS (CS-23.1329 + 23.779 and AC 23-17C, 23.1329 refer)

LAA/IC-APY
Issue 2

A/C Type:

Reg:

LAA Mod No.

Date:

Auto-pilot (Yaw) Make/Model:

1. Check the force required on the rudder pedals for the pilot to overcome the yaw servo. It must be possible to quickly and positively overcome the servo. The force must be adjusted to the minimum value possible to prevent servo clutch-slip when the auto-pilot maximum yaw rate command is activated.

SAT	UNSAT	Comments:
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2. Check that each manual control and switch for the auto-pilot (yaw) system is easily accessible to the pilot in flight. Check that each control operates in the natural sense and that each control is marked as to function and sense.

SAT	UNSAT	Comments:
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3. A switch must be provided to quickly disengage the autopilot in flight e.g. in the event of a malfunction. In the case of a dual control aeroplane, a disengage switch must be easily accessible to both pilots in flight.

SAT	UNSAT	Comments:
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4. If the maximum force required to overpower the servo is greater than that which can comfortably be applied by the pilot, the disengage switch(es) for the auto-pilot (yaw) system must be mounted on the stick grip(s).

SAT	UNSAT	Comments:
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5. An alternative control to that specified in 3. and 4. above must be provided, e.g. a clearly marked and easily accessible circuit breaker.

SAT	UNSAT	Comments:
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6. For auto-pilots designed to command servo disengagement following pilot applied force to the controls, an aural 'auto-pilot disengaged' warning feature must be activated. Record the type of warning (horn, voice, etc.) and whether through the headphones, speaker, horn, etc.

SAT	UNSAT	Comments:
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7. Check that the maximum servo travel either in normal use or following a malfunction (servo runaway) cannot cause undue loads or travel in the control system or the control stops or cause over-centre geometric lock which could jam the system.

SAT	UNSAT	Comments:
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8. There must be a clear indication (e.g. a warning light) showing the selected mode of operation. Selector switch position is not by itself adequate. In the case of a dual control aeroplane, the indication must be clearly visible to both pilots.

SAT	UNSAT	Comments:
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9. Check that each part of the linkage connecting the servo permanently to the flight control system complies with normal requirements for flying control systems with regard to integrity, locking of bolts, strength, freedom from possibility of jamming etc. – specifically, ensure that the servo arm/pulley retaining bolts are installed using thread locking compound unless another acceptable bolt retaining method is employed.

SAT	UNSAT	Comments:
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10. Check the rudder moves in the correct sense and with the correct range of movement stop to stop.

SAT	UNSAT	Comments:
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11. Check the possibility of the servo being coupled up the wrong way round during maintenance is remote (e.g. it should not be possible to install the servo arm at 180 degrees to its normal position and electrical plugs and sockets should be suitably polarised to ensure correct orientation).

SAT	UNSAT	Comments:
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12. Placard to be fitted clearly in view of the pilot, stating:
'ENGAGEMENT OF AUTO-PILOT BELOW 1000 FT AGL PROHIBITED'.

SAT	UNSAT	Comments:
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I certify that the above checks have been carried out to my satisfaction.

Name:	Signed:	Insp. No.:	Date:
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Note: **Valid flight test authorisation issued by LAA Engineering required prior to flight.**